

AVIATION WEEK

AUG. 11, 1952

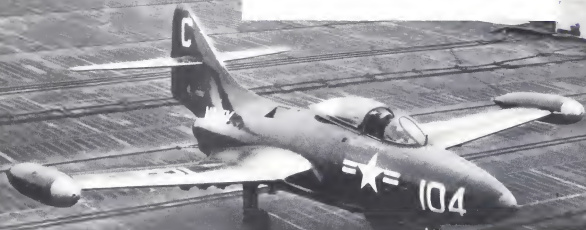
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50 CENTS



Landing number 39,000 for the first in the fight!

It was no coincidence that the 39,000th plane to come to a stop in the U. S. S. Midway's arresting gear was a F9F PANTHER. These battle-proved fighters, first Navy jets to see combat in Korea, have been taking off and landing on this big carrier's deck for over two years. That the once spectacular is now the commonplace reflects Navy and Marine Corps skill and teamwork . . . plus the inherent ruggedness and dependability of the GRUMMAN PANTHER.



GRUMMAN AIRCRAFT ENGINEERING CORPORATION, BETHPAUL, MICHIGAN

Contractors to the Armed Forces

"EYES" of the SCORPION



demand constant frequency

AC Power*



"PACKAGE-TYPE"

CONSTANT SPEED DRIVE
MOUNTS DIRECTLY TO ENGINE

On the Northrop "Scorpion," the Sundstrand "Constant Speed Drive" mounts directly onto the engine. Other Sundstrand drives available include the "integral-type" designed into the main aircraft engine, and the "split drive" type where the hydraulic drive is connected to the engine accessory port, while the hydraulic motor is integrated to the generator, usually located in the tailroom.



**SUNDSTRAND
AIRCRAFT
HYDRAULICS**

SUNDSTRAND WORKING PUMP CO.
ROCKFORD DIVISION, ROCKFORD, ILL.

*SUNDSTRAND'S constant speed drive PROVIDES it!

Hundreds of electrical components in the elaborate radio search gear of the U. S. Air Force's latest all weather interceptor fighter—the Northrop Scorpion P-83—greatly increased the demand for a dependable source of power. The answer—constant frequency AC power made possible by Sundstrand's Constant Speed Drive. Sundstrand's Drive maintains the varying speed of the turbo-jets to constant speed for driving the AC generators.

The result is a dependable source for constant frequency AC power and a divided saving of both weight and space in the aircraft—especially important in the long range "Scorpion" which travels at speeds in the 600 MPH class and at altitudes over 50,000 feet.

If you have an electrical problem, call on Sundstrand's radio research, expert engineering, precision production for help.

B.F. Goodrich



Wafer-thin rubber sandwich solves icing problem

THE NEW JET ENGINE has only a few more jobs to provide extra power when the Convair B-58 takes off, in power-driven slush and for that extra burst of speed needed over the runway. The aim of the aim, a job to be covered or kept free from icing.

That's the reason for the slush-like "doors" you see. Doors that must open when the extra power is needed. And ice forming on flight could seal the doors tight. Here had to be provided, you the slush had to be slush water then. The manufacture of the doors though he could do it by making the slush like a sandwich—of the sand-

with filter could be made thin enough and still provide the amount of heat needed to keep off ice.

The experience of B. F. Goodrich with hundreds of airplane wing pack-lenses came in handy on this one. It took some precise engineering to solve the tough problem of slush, but it was done. The doors that would the trick is only 1/4 of an inch thick. The use of precision work is embedded by a unique BFG method into a layer of Fibreglass impregnated with rubber-like material. It provides all the anti-icing heat needed to keep the doors free for all slush.

B. F. Goodrich offers the aviation industry a background of almost 25 years experience in anti-icing problems working with both test and production. Dozens of BFG products for anti-icing include: wires, shields and liners, Plastick adhesives, Precision Sealing Zippers, hot cells, Kinetics, actuators. The B. F. Goodrich Company, Aviation Division, Akron, Ohio.

B.F. Goodrich
FIRST IN RUBBER



Are you taking full advantage of the constantly growing range of forgings? Typical is this aluminum alloy forging with a projected area of more than 1,000 square inches used in the wing structure of a modern military bomber. Such forgings are today made possible by the use of the largest die forging press in America (18,000 tons). For hammer or press die forgings of aluminum, magnesium or steel, Wyman-Gordon engineers are ready to serve you—there is no substitute for Wyman-Gordon experience.

Standard of the Industry for More Than Sixty Years

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FORGINGS OF ALUMINUM • MAGNESIUM • STEEL
WORCESTER, MASSACHUSETTS
HARVEY, ILLINOIS DETROIT, MICHIGAN

NEWS DIGEST

Domestic

Douglas B1-B6 test jet engine components bomber will be built for USAF Tactical Air Command at the company's Long Beach division. Shows the bomber 600 500 mph, then place them from May, across land AID, Long Beach test will compare the design to USAF requirements.

Jet plane utilization was on by USAF pilots flying a Lockheed T-33 trainer 400 by 16 min. during July, north divide the previous U. S. record for single-engine jet of 202 by. On the final day of the record, pilots kept the T-33 aloft for 23 by 27 min.

Malvin J. Mann, former Navy's congressman, noted for his support of aviation, acted from U. S. Marine Corps as major general. World War II USMC pilot Mann has earned blood stars.

Gold aircraft shipments during July totaled 150 planes valued at \$21.4 million and airplanes \$50,780 in aircraft weight. During May 524 civilian aircraft valued at \$1.6 million were shipped.

National Air Races for 1951 will be held at Dayton's Municipal Airport, Yonah, Ohio, in connection with Dayton's observance of being's golden anniversary. Date will be Sept. 7-7.

Drafts court order restoring David E. Belknap to presidency of Air Line Pilots Assn. has been temporarily set aside as new ruling by Federal Appeals Court, Denver, IF, which also set aside the appointment of a manager for the union's financial affairs.

Belknap, Gen. Arthur E. Buehler, (OK), was based in Arlington County July 18. He was the only observer to win the sitting of seat during World War I.

Stanley Spaulding, first Secretary of USAF, then head of Reconstruction Finance Corp., suggested a land-based victory in Missouri Democratic primary for nomination as senator over his opponent, Missouri Attorney General J. E. Fisher, President Truman's candidate.

Major Gen. Follet Bradley, USAF (Ret), 62, died Aug. 4 in March N. Y. He had been assistant to the president of Sperry Corporation Co., since retirement from USAF in 1942. Bradley's flying experience dated back to 1912.



MERGE AGREEMENT joining Bell Telephone Assn. and Bell-Columbia Airlines is signed by E. S. Russell (right), with A. W. Miller, NECA president looking on. Approved by CAB, recently passed by

majority of both concern stockholders, the merger issues largely whether the work. The merger move by the carrier more than double Bell's domestic route mileage.

Financial

Boeing Airplane Co., Seattle, Wash., reports net earnings of \$1,677,675 for the six months ending June 30 in after and after income of \$399,761,874. Its total orders total approximately \$1.2 billion, not including contracts under negotiation.

Republic Aviation Corp., Farmingdale, N. Y., had net income of \$2,626, 125 for the first half of 1952 after provision for taxes. Sales for the period, ended June 30, totaled \$194,256,855. Republic's backlog is estimated at \$931, 154,788.

Bell Aircraft Corp., Bethel, N. Y., reports \$294,184 profit after taxes on sales of \$58,516,145 for the first half of 1952. Its comparable second half year, Bell's profit was \$782,861. Backlog at June was \$420 million.

Ryan Aeronautical Co., San Diego, has declared a regular quarterly dividend of 12 cents per common share payable Sept. 12 to holders of record on Aug. 22.

Continental Air Lines net income for first six months of this year was \$85,946 in total operating revenue of \$4,930, 257.

Consolidated Value Aircraft Corp., San Diego, Calif., reports net income of \$4,072,136 for the first half of 1952 after provision for federal taxes. Total sales during this period were \$113,

\$15,795. Company's unfilled orders are estimated at more than \$1 billion.

Mid-Columbia Airlines has declared a 25-cent dividend payable Aug. 13 to stockholders of record Aug. 1.

North American Aviation, Inc., Inglewood, Calif., reports net income of \$5.2 million for the nine months ended June 30 on sales and other income totaling \$294,129,069. NAA's backlog is estimated at more than \$741 million.

Curtis-Wright Corp., and related assets, report net profit of \$1,137,108 for the six months ended June 30. Total net sales for the period came to \$144, 548,561. As of June 30, C.W.'s unfilled orders exceeded \$1,010,800,000.

Northwest Airlines reports first passenger revenues of \$4,245,118, a monthly record for the company.

Grumman Aircraft Engineering Corp., Bethel, N. Y., had a net income of \$2,065,329 after taxes for the six months ending June 30.

International

Japan Air Lines reportedly has ordered two DC-6A (series II) (Boeing-Boeing) at cost of \$1.1 million.

Sabena Belgian Airlines has ordered two Douglas DC-6A Liberator freighters for delivery by August, 1954. Power plants will be the PW-6 (3,200-2813).

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MACWHYTE Aircraft Cable, Fittings, Assemblies



Macwhyte "Hi-Torque" Aircraft Cable has maximum tensile strength and exceptional resistance to wear and bending fatigue. Being properly PATENTED, it lays down with its tendency to twist or coil which provides better cable installation at lower cost.

You can order Macwhyte "Hi-Torque" Aircraft Cable in steel box, specified lengths, or assemblies. Macwhyte "Bolt-Link" and "Socket-type" terminals are applied loose or attached to cable.

Macwhyte Aircraft Casting A-3 is available on request.

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Manufacturers of "Hi-Torque" Aircraft Cable • "Bolt-Link" Cable Terminals • Cable Assemblies • Tie Rods • Braided Wire Rope Rings • Springs, Cotterpins, Stainless Steel and Mild Steel Wire Rope



Member A.S.M.E. and A.S.A.
"Hi-Torque" is a registered trademark

WHO'S WHERE

In the Front Office

George R. Shaw has been made vice president of Allison Aircraft Division of Allison Corp., Indianapolis, IN, and Donald W. Shaw has been appointed vice president industrial relations. Shaw joined the firm in 1947, during the war, served with ANF on the Eastern District Army Airfield, and then moved to Allison Aircraft Division. Shaw has been with Allison for 13 years, since 1951 has been industrial representative of the Allison Committee in an advisory capacity to WSO.

E. Swain Rouse has been made president of Warner Corp. division of The Warner Corp., Memphis, Tenn., succeeding A. F. Rouse, who retired from active management of the division after 21 years of service. He last night as president. Vice president Warner Corp. division vice president is E. J. Auld, William H. Corwin, John C. O'Neil and Andrew W. Ross.

Sam C. Mansfield, manufacturing manager at Lockheed's Golden division since 1953, has been named vice president manufacturing manager for the firm. He joined Lockheed in 1946, previously was with Douglas Aircraft Co. for 17 years.

Douglas F. Johnson has been named president of the newly reorganized Tulsa Trading Corp., Tulsa, Okla., "airline" subsidiary, in a move to acquire a number of small property and construction divisions of Tulsa. Other appointments for Tulsa: David M. Nelson, chairman of the board; E. H. Day executive vice president; Allen A. Rouse, Eastern vice president; Michael J. Nease, Western treasurer.

Joseph A. Fench, executive public relations director for Northwest Airlines, has been designated vice president of Travelodge Industries, Inc., New Richmond, Wis., and Portland, Ore., a holder of plane products and airplanes and engaged in produce and real estate. He will be in charge of the firm's public relations program.

Changes

Joe Matheson has been designated assistant director of contracts for sales in Bell Aircraft Corp.'s Helicopter division, Ft. Worth, Tex., and G. E. Bell has been made assistant director of products administration. S. Gottlieb has been promoted to supervisor of military contracts. Tom Shaw is now contract administrator for domestic sales and Jack Bays has been made contract administrator for export sales. Jack Keller has been made supervisor of mechanical sales. John Shaw has been appointed contract administrator for sales in Germany and Robert Kaufman, contract administrator for military contracts.

Charles D. Adams has been named public relations director in charge of advertising for Deere Helicopters, Inc., Dubuque, Iowa.

Robert J. Dillman has been named sales manager for International Division. Allen R. DeWald, Lubbock, Tex., is Chief N. J. (Continued on page 11)

INDUSTRY OBSERVER

Latest Maritime Board aircraft schedule indicates that powerplant of the F-4B-100s from Wright J65 to General Electric J71 and Douglas B-65 (A20) engine installations includes from Westinghouse J60 to Allison J71, Westinghouse F-88F also includes J71 installation in place of the Allison J81 engine powerplant of the F-88C. Both the J71 and J71 are rated at 10,000 BHP thrust.

Being prepared for immediate usage of the B-52 right-hand engine overhaul production, currently scheduled for next year, in 12 per month has been done by USAF as "two early." Apparently, Pentagon sources report, USAF will not pursue B-52 production was heavily until detailed engineering studies of two separate bomber projects are thoroughly evaluated.

Military sources forecast outlook on the near future of jet engines developing less than 7,500 BHP thrust. Current aircraft engine schedules indicate follow-on production only of engines based on unimproved thrust of 7,500 BHP and over. All of these, it is reported, are going to get even more power with afterburners.

Lockheed Aircraft is under consideration in a second source supplier of the General Electric J71-102 interceptors. Plans will be provided by Pratt & Whitney J65, with electronic control system furnished by Hughes Aircraft. Originally nicknamed "54 Intercepter," difficulties in electronic pilot system development has forced rescheduling program "55 intercepter" (Aviation Week Nov. 3, p. 11).

According to engineering schedules, Republic Aviation's response to the AF-102 will not be ready for flight test until late 1954. Designed for much the same mission as the General Electric J71-102, considerable experience of F-403 components is being gained in test flights of the Republic NF-50 now under way at Edwards AFB, Calif. F-403 interceptors, among other design innovations, access target wings, and complete powerplant installations with both jet engine and rocket nozzles.

Bids were received in Washington last week for 240 training light trainers for use by the new USAF rocket contract pilot training schools. Specifications for the new trainer which was prepared by USAF include installation of a 90-hp, piston engine, (presumably) training engine, a Continental D-1200 in the cockpit for the instructor and a left to right seat for the student. All Force sources report that specifications generally are comparable to the T-38C-15, developed in 1950. Contract will total approximately \$716,000.

Federation Aeronautique Internationale has confirmed new world record for Category B aircraft (3,000-3,500 lb.) established by Max Caouette flying a stock Piper Four powered by a Lycoming 125-hp engine. Caouette set a distance of 2,462 statute miles between Los Angeles and New York, doing a 34.34-hr. average flight, May 1. The plane carried a total of 135 gal. of fuel. Average speed was 99.19 mph.

One problem reported for the Lockheed F-94 intercepter series is an extreme yawing condition. Pilots report considerable difficulty in holding line on course in order tracking before and after target lock-on.

Ford Aircraft Engine Division, Chicago, has disclosed that the first General RB-101, powered by an Ford-built RB-101-01 Pratt & Whitney-designed engine plus the usual four General Electric J47 turbojets, has been test flown successfully to a 40,000 ft. altitude from Ft. Worth.

Helicopter manufacturers can supply most commercial customers after 1954. Washington sources report, due to quality buying of rotary aircraft by all three military services. Industry sources predict an upsurge of helicopter production in the part of major orders as a result of the military desire to achieve helicopter production to civil units.

Washington Roundup

The Platforms

The latest campaign of Presidential aspirants for an all-powerful air arm soon reached the preflight test of either the Decascope or Rembrandt plane platform designed at Douglas.

• **The Rembrandt** plank called for "the quickest possible development of appropriate and completely adequate air power." It flew out on its carrier wing and ground forces on a criss-cross strata. Adequate defense, the platform reads, requires "the continuous readiness of coordinated air, land, and sea forces, with all necessary installations, bases, supplies and munitions, including atomic weapons, in abundance."

• "Her power" was not once mentioned in the Decascope platform. The catch aspect of "decade" angled out for specific confidence and support was aggressive. "The Decascope plan, certified 'Defense Needs' pledged 'We will continue to recognize aggression as a central defense industry, and to make in providing of the necessary tools, machinery, facilities, and supplies needed for farmers in meeting production goals'."

Significance of the platform from the defense view is the different emphasis.

• **The Decascope** stressed "collective security" consolidating the military strength of non-Communist nations and making military aid to allies.

• **The Rembrandt** put emphasis on strong U. S. armed forces.

The Candidates

Gen. Douglas Eisenhower, top Air Force officer report, is not at the front line area who proved the possibilities of air power long since his struggling days, and still retains that grip.

But Eisenhower lacked in Defense Secretary Louis Johnson's general theme, in the fall of 1949, that U. S. military strength should be subservient to the country's economic health. However, after Johnson spent \$1.5 billion in a \$15 billion budget for the military services that went to Congress each, in 1950, Eisenhower pointed out that the weakening was going too far, particularly in an emergency. The result, Eisenhower and Johnson agreed on \$150 million more for aircraft procurement, which Congress granted.

Gen. David Stevenson's knowledge of military defense is well made.

During World War II, he studied defense and strategy in both theaters, first as special assistant to the late Secretary of the Navy Frank Knox and then as a member of the War Department interim that went to Europe in 1941.

Sen. Richard Nixon's four years service with Naval Aviation during World War II included Aviation aide-instruction at Quantico, Va., command of the South Pacific Command Air Transport Command at Guadalcanal, later at Green Island, general representative for Walter's first term in the Senate as a member of the Senate.

Sen. John Sparkman is a leader of the Senate, he is pushed for more competition in air transportation and has backed the scheduled expansion of the industry on annual safety security legislation.

A friend of the Air Force, Sparkman voted for the 70-year program back in 1949, when only nine Senators did.

Seaplane Bamber?

The Navy again is considering the possibilities of the seaplane as a bomber.

Its advantage over a carrier-type "attack plane" is long range.

The industry for engineering designs of a seaplane comparable to the B-47, programmed to become the best, began of USAF's long-range strategic force, and will be ready to implement them by 1975 if the Navy wish.

Assistant Secretary of Navy for Air John T. Johnson recently "We are looking into all seaplane possibilities with an interested eye. This would be an additional capability, not a substitute capability, for carrier aviation."

"The seaplane force wouldn't have the mobility of the carrier force, but it would have more flexibility than land-based air."

Navy's seaplane fighter-like Corsair 277-will track the carrier of the seaplane into Naval air's striking area. It will make its first flight, probably, this fall.

Curbback in Spares?

Watch for a major curbback in Air Force's program for parts, equipment, and engine spares.

Congress already has indicated that if USAF doesn't reduce the percentage of aircraft money that goes into maintenance, it will be withheld in the future. Senate Appropriations Committee voted to slash \$600 million off the fiscal year's money, allocated for spares. It was later explained to give USAF time to work out a new policy.

Sub-four percent of the money for each USAF plane goes into spares.

Sen. Henry Reagan, a member of the Appropriations Committee, observed: "This means that for over 100 new airplanes turned over to the Air Force and placed on an air field, the equivalent of 64 planes is stored in the form of spare parts."

USAF's Undersecretary Bernard C. Galtier responded: "We have to cut that 1-40 cut not just once, once a long period of time, as one nation a force like this and have that crash of one nation going into the past."

Because a challenge to USAF. The 145 wing, located in the Pacific, the Joint Chiefs of Staff's target date of mid-1954 (end of the "strategic" date of mid-1955), it alone of the 54 billion marked the space that was used for complete aircraft.

Pressure to end look the space program was set off by testimony that during one of an report on the next two years will have USAF with surplus space in storage and ring over \$120 million.

Army Aviation: A New Sponsor

Army aviation will be approached quickly when it is established from the Ordnance Corps and turned over to the Transportation Corps. The shift, now underway, will be completed around the first of the year.

"We live with," Ordnance has served in an off-and-on middle-man between operations? Army commands, determining plane requirements, and Air Force doing the actual procurement.

Transportation Corps wants to exploit both the helicopter and fixed wing for its operational functions.

It wants to do a wide variety of missions, directly, with USAF chartered as a go-between.

—Katherine Johnson

AVIATION WEEK

Douglas Ready to Build Jet Transport

- Plans now firm; mockup being shown to airlines.
- Company to invest up to \$40 million.

Douglas DC-8

(Tentative Specifications)

Span	127 ft.
Length	115 ft.
Wingspan diameter	150-150 ft.
Gross weight (design range)	220,000 lb.
Gross weight (maximum)	180,000 lb.
Cruising speed (40,000 ft.)	560 mph
Stall speed (maximum)	2,500 ft.
Passenger	70-750
Crew	5
Powerplant	4 F4W J57
Thrust, each	15,000 lb.
Cost (production model)	\$3 million
Cost (prototype)	\$30-44 million
First flight (production model)	1918

Douglas Aircraft Co. has spent to all work into the race to build and sell an American commercial jet transport.

Douglas is now submitting a full-scale aircraft mockup of the new jet to the FAA in Santa Monica to meet prospective airline customers. Top executives have made a first policy decision to move full speed ahead with a powerful, financed program aimed at capturing and dominating the jet transport market in the same manner that Douglas has led the piston-powered transport field since the days of the first DC-3.

Private Venture-Sources close to the Douglas project estimate that from \$30 to \$40 million may be required to launch adequately the DC-8. Douglas executives have made the decision to support the program without any government subsidy, and to build the aircraft in a purely commercial project, as the world's largest aircraft manufacturer.

Basic design philosophy of the DC-8 is that it must be an aircraft, not a truck-like transport that will enable airlines to operate it at a profit. This contrasts sharply with the approach of some Douglas competitors in concept.

USA's Undersecretary of the Air Force, Douglas, says the aircraft design is a commercial one.

Although Douglas officials are quick to admit that there will be some configuration changes as the DC-8 is developed, they say the design calls for a low-wing, single-engine, with a wingspan of about 115 ft.

Concepts will be far advanced versions of the Pratt & Whitney Aircraft JT7 Turbofan engine, with a thrust of 15,000 lb. thrust, class. The aircraft has a low-wing, single-engine, with a wingspan of about 115 ft. The aircraft has a low-wing, single-engine, with a wingspan of about 115 ft. The aircraft has a low-wing, single-engine, with a wingspan of about 115 ft.

Specifications—Wing span is about 115 ft. with a wingspan length of about 115 ft. A single jet engine also incorporates 35 deg sweepback.

Two versions of the DC-8 are planned. One will gross about 150,000 lb. for domestic airline operations and the other will have a 220,000 lb. gross weight at long-range operations. The domestic version is expected to have a cruising range of 3,500 mi. with fuel burners for 300 or additional and one hour of holding at 15,000 ft.

A cruising speed of 560 mph at 40,000 ft. is expected. The aircraft has a cruising range of 3,500 mi. with fuel burners for 300 or additional and one hour of holding at 15,000 ft.

Two seating arrangements will be available—a 70-90 seat interior for first class service and 120 seats for high-density cargo operations. Fuselage diameters of the DC-8 is still under debate with 130 and 150 in. various under consideration. The larger diameter would permit an almost seating in the cabin.

The cockpit configuration has not been finalized but previous work probably be made for a single engineer in addition to pilot, copilot, and navigator. Landing gear is a four-wheel bogie type located ahead from the engine pods. It will field in wide use the fuselage and wing root.

William Leach-Douglas doesn't take the gamble of a possible financial commercial jet transport project has given it a clear-cut lead over its traditional competitors in the commercial market. Lockheed and Boeing, Lockheed has recently abandoned previous designs of a four-jet transport based on the general configuration of

the F-92 supersonic prototype fighter and is working on a new design approach to the problem.

One major vision in prospect is the Lockheed design in reconfiguring the engine. The four jets were clustered in the aft section of the fuselage in the earlier Lockheed design, primarily to keep the passengers forward of all engine noise.

Boeing has not yet gone beyond preliminary design work on a number of jet transport proposals.

The DC-8 is expected by Douglas to be the standard jet transport for the next decade since it will have an extremely high altitude design speed (Mach 3.0). Power requirements to push transports into the transonic range are not likely to be available economically by at least 10 years.

Cost of production models of the DC-8 is estimated at about \$5 million apiece. First production model is scheduled to fly in 1958.

In addition to costs of leading U. S. airlines management and operations executives in Santa Monica, Douglas also engineers have been touring Europe, planning the DC-8 among foreign airlines.

There is a large backlog and a continuing flow of orders for piston-powered Douglas and Lockheed transport planes from both domestic and foreign airlines is interpreted by Douglas as indication that the large-scale, profitable world airline market for jet equipment will be open about the same time the DC-8 becomes available.

down by government subsidization appears it failed to be achieved.

North American and UAW-GEO agreed July 30 to voluntary arbitration, that averting a strike of the company's 28,000 employees at plants in Ingersoll, Downey and Tucson, Calif., and Columbus, O. The union wanted, as a condition of arbitration, that the firms plant wages be covered by the national agreement. UAW-GEO initially was negotiating rights at Ingersoll.

Both sides agreed that 12 cents of a 15-cent cost-of-living bonus will be put into the base hourly wage rates, thus raising the minimum from 12 cents. The arbitration panel will decide only the question of a general, across-the-board wage increase. UAW-GEO has demanded an increase of 17 cents an hour. The company has offered five cents.

F-66H Production Due at Columbus

Despite recommendations by Aircraft Production Board Acting Chairman W. L. Campbell (Aircraft Week, July 25, p. 12) to put over the North American F-66H Sabre fighters in favor of the later North American development, the XP 109 Sabre 41, production will start at Columbus on the F-66H late this year. Indications are the XP 109 will not be used for production for some time to come.

North American announced that the first two models of the 10 series were being built at Los Angeles but that subsequent production would go in the Columbus North American plant. It will supersede the earlier F-56, now being produced there.



BRITISH MISSILE BLASTS OFF

At an experimental station "somewhere in England," play of booster starts light up a fluidic guided missile as it leaves its

The F-66H will be slightly larger than earlier Sabres, and is powered with a General Electric J47-GE-19 engine, more powerful than those used in earlier F-56s.

Other improvements include a new suspension and release mechanism for carrying wing tanks in conjunction with bombs or rockets, streamlined type canopy similar to F-48D, an improved cockpit to give pilot more vision and vision, improved pilot ejection seat mechanism, larger horizontal tail surface, canopy deflated found in earlier models, beefed up landing gear. Like other late model Sabres, F-66H will have positive boost controls on its entire horizontal tail surface.

USAF on Lookout For Irregularities

The tremendous jump in AMC procurement personnel—from 4,000 persons to 12,000 shortly after Korea—and the considerable business almost buying and contracting employees, offer the possibility of purchase irregularities, Air Force leaders.

Six cases of irregularities, ranging from direct bribery to use of improper influence, were reported by USAF at a recent congressional hearing.

The total loss value of the contracts involved on the six cases was \$14,565,856—about one-third of one percent of all procurement initiated in fiscal 1951. A large part of the dollar losses involved are expected to be recovered, leaving a significant loss to the government at \$4,565,856. No delivery of subcontracted material was made nor was any needed material irregularly delayed as a result of the irregularities reported. AF stated

While disclosing employees in working irregularities, Air Force told the congressional group it "knew that neither the best organizations, nor the best individuals, nor the soundest procedures, nor all of these together, will totally eliminate irregularities."

But AF hopes that by setting up "proper organizations, policies, procedures and practices," it will be able to keep such cases to an undesirable minimum.

Junkies Returns

(McGraw-Hill World News)

Frankfurt—Typical of arranging German aviation is formation of Junkers Flugzeug- und Motorenwerke GmbH in Kassel with an initial capitalization of 20,000 Deutsche marks (\$4,761). One of the company's business managers is Kurt Adenauer, a nephew of Germany's Chancellor. Frank Adenauer has been chosen as chairman, says Kassel. The firm plans to work on development, production and sales and repair of planes, engines, parts and aviation mechanical tools within framework of existing regulations.

WHO'S WHERE

(continued from p. 21)

and Melvin G. Scheraga has been promoted to assistant industrial sales manager of the firm.

D. Marshall Kiehl has been appointed manager of commercial aircraft sales for G. M. Gorman & Co., Pasadena, Calif., sales of passenger aircraft, and consists for aircraft models and military planes.

Dr. C. J. Barreiros has been promoted to director of engineering for E. M. McElroy & Co., Indianapolis, Ind. He is known for his work on guided missile and aircraft component and control systems.

Robert G. Sullivan, formerly general manager and sales for Curbish-Town & Sons Corp., has joined Southern Aircraft Services, Springfield, N. Y., as a staff assistant to the executive vice president, where he will handle long range sales activities in new aviation developments.

John A. Poon has been named manager of purchasing for Aviation Co. Textile division of Westinghouse Electric Corp., Philadelphia, Pa. William C. Wilson has been named purchasing agent for the South Philadelphia plant and George A. Feller now is purchasing agent for the Kansas City plant. E. K. Nay has been made assistant to the manager of the Westinghouse's Baltimore division, and will be responsible for uniform manufacturing policies between the Electronics, Aviation and V. R. divisions.

Robert Arroy, formerly sales manager for Mel Arroy's Dallas Textiles, has become public relations and advertising manager for the Indianapolis-based head quarters in Indiana.



JOY AXIVANE® AIRCRAFT FANS

warm airborne troops before take-off

To protect our airborne troops in frigid areas before take-off, Joy AXIVANE Aircraft Fans are installed in these large troop-carriers to blow heated air into the main cabin while the plane is on the ground. Heat effect is utilized for this purpose after the carrier is airborne. Air from the fan is mixed with a measured amount of air from the heater to provide the desired air temperature in the cabin. Then, cold weather is no hindrance to the fast, efficient transportation of our fighting men to any theatre.

This highly-efficient L-5 H.P. fan produces 1100 C.F.M. at 5.5" static pressure, yet weighs only 22 pounds and is only 3" in diameter. A & N design specifications. Superior features of all Joy Aircraft Fans are compact design, shock-resistant strength, minimum operating noise, and the most favorable air volume-to-weight and electric-to-air power ratios.

Joy designed and built each fan to fit exact requirements for which it is intended. Each fan, therefore, is custom-engineered for highest efficiency. For many purposes such fans can be supplied from the warehouse line already designed. Each single and two-stage unit is available. Optional features include variable air flow control, heated or dehumidified air, and motor. Motors, speedometers, and control systems were required.

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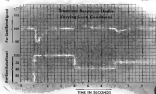


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AERONAUTICAL ENGINEERING



B-36 PRESSURIZATION and air conditioning meet challenges of 100-deg. temperature change and 40,000-ft. altitude change.

What High-Flying B-36 Taught Convair

High-altitude safety and comfort are interrelated; they must be integrated at the design level.

With military planes demanded to fly even higher than their present ceilings and the commercial jet transport just at the beginning of its high-altitude flight, cabin pressurization and air conditioning take on greater significance.

As usual, the military vehicles have been the proven for high-altitude techniques such as these, but little information has been made available.

B-36 Experience—Knowledge, however, is sought in the engineering and mechanical aspects of pressurization and air conditioning at today's high-flying planes was revealed by W. C. Dietz of Consolidated Vultee Aircraft Corp.'s Ft. Worth staff. In his paper, "High Altitude B-36 Cabin Pressurization Operating Experience," presented before the Society of Automotive Engineers' National Aeronautics Meeting in New York, Dietz points out that design is an up again.

Cabin pressurization and air conditioning are so related that successful operation of either system depends on a complete integration of the design level, says Dietz. The air source must not only be adequate to maintain pressurization, but must provide sufficient air for ventilation heating and cooling. From a design standpoint a large portion of the engineering field is concerned.

Military vs. Commercial—Solutions to problems with the B-36 are not necessarily acceptable solutions for a commercial airplane. There is, however, a common objective—safe and comfortable altitude operation.

In the case of the B-36 the pressure rooms for flying at high altitudes are critical to commercial aircraft operation. In high-altitude flight it is possible for exposed passenger comfort and survival.

There are, however, a number of problems associated with commercial operation at altitudes above these two commercial levels. Major problem is that of safety of cabin pressure is lost. For a transient and projected case in good physical condition, as in the ordinary service, and with adequate oxygen provisions, the danger is minimized.

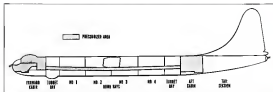
In the case of commercial operation, involving children, aged people, and those with physical impairments, loss of cabin pressure can be a potentially dangerous—possibly fatal, Dietz explains.

Temperature—Ventilation—Light in the high-altitude range has presented a number of new problems in addition to those directly associated with the increased pressure and the physiological factors involved in an engine deoxygenation. Among these are temperature extremes encountered.

In several instances a B-36 flight was diverted to the ground with an ambient temperature of 100°F, but at altitudes of 40,000 ft., -100°F was encountered. This extreme variation requires the utmost from the heating and heating system along with a control system adaptable, and with adequate response, to this extreme range of operation, Dietz continues.

To add further complexities to the demands on the heating and cooling system, the effects of heating in other industries are quite marked at high altitudes.

Comfort—Conditioning—Yet another consideration is the necessity for providing adequate crew and passenger



PRESSURIZED AREAS of the Convair B-36 bomber are the forward and aft cabins and the long tunnel connecting them.

combat facilities. The combat factor is a more important one due to the longer range and consequent longer duration flights that will be flown with operation at higher altitudes.

Providing adequate passenger room, first but at times, always, has been an important factor in conventional aircraft design, and much progress has been made along this line. However, in military tactical aircraft, comfort has in the past been of somewhat secondary importance. Experienced with the B-36 has pointed out the vital necessity for improving crew comfort in the elimination of fatigue for more efficient utilization of the flight crew's talent, Datta says.

• **Oxygen Debris**—From an engineering viewpoint it is necessary to understand the location of the human body in the most important factors which affect the safety of passengers, and come in the event of an explosive decompression. These are the two primary problems, says Datta, and one of the most important factors in the design of the aircraft is the safety of passengers.

It would appear important at the present state of development of oxygen equipment to protect passengers properly in the use of this equipment to eliminate danger from flames, he says—the only solution that would appear feasible at this time being a rapid descent to about 10,000 ft. Estimates are that this would have to be accomplished in 4 to 5 min.

While some physical injury due to the rapid increase in pressure with decrease in altitude might occur, it would be considerably less dangerous than a loss of control due to the loss of altitude, Datta says. The possibility of "bends" or decompression is not just because a descent rate rapid enough to prevent nausea will also cause the danger.

• **Passenger Aspects**—The second problem—passengers, or crew being in close proximity to areas of failure—obviously cannot be overcome by location of the

passengers and crew if efficient space utilization is to be realized, because a structural failure could conceivably occur at any point, Datta says.

However, if consideration is given to the structural design to prevent passengers, type structural failures, the problem can be confined to the most probable items of failure, such as wing doors and doors which, if failure should occur, would cause physical injury by collision with the surrounding structure or ejection from the airplane due to the rapid flow of air through the opening.

After a window was lost in the B-36 and a crew member was ejected, several methods were considered to prevent a recurrence of the accident, Datta relates. The possibilities investigated were the use of safety harnesses, nets and in some cases pressure harness.

Safety harness have proven satisfactory in some modifications, but they present a restriction to normal crew movement and are not considered as good as a secondary positive barrier.

In this case an intact safety harness would be useful in preventing the cabin pressure but not completely prevent ejection. This design, while it will not prevent depressurization of the cabin, will increase the time for decompression and eliminate the possibility of ejection.

• **Rapid Descent**—The possibility of a rapid physical injury as a result of the exposure of internal body parts during decompression in aircraft of the B-36 or conventional type is not prohibitive. The opening, such as the window and doors which are the most probable cause of failure, are sufficiently small in relation to the cabin volume that the time elements of decompression are long enough to provide any danger from this source, Datta states.

As a typical example, applying Air Force Armory Medical Laboratory data to the forward cabin of the B-36, the damage which could be sustained

without exceeding the safe limits of relative gas exposures is approximately an opening of 55 sq ft. A structural failure which would result in this much of an opening would be of a major nature, and extremely unlikely.

• **Maple, of Safety Department**—to the safety of high-altitude pressurized operation is an adequate structural design of the pressurized components.

For military tactical aircraft, however, by experienced crews, the danger of a decompression are minimized.

• **Sufficient oxygen provisions** are provided for the possibility of damage to the pressurized components in ground or in-flight.

• **Because of the safety provided by oxygen equipment and crew training**, the structural design of safety for a tactical aircraft design are not in critical condition.

• **Passenger Failure**—In addition to providing ample oxygen equipment, the primary structural components, an important factor is the prevention of passenger type failures.

Because the present limits are not adequately realized in cases of major structural failures, there are possibilities of passenger failures due to overloading of passenger failures due to overloading of the aircraft structure. Datta says there is very little data available on the stress level at which structural elements, such as fuselage joints, can be safely operated in relation to the use of damaged parts without a progressive type failure and resulting explosive decompression.

Maple says, and extremely in previous structural specifications on the B-36, was not considered acceptable for use in the pressurized cabin due to its lower ductility and high notch sensitivity and, therefore, its questionable ability to withstand battle damage. For this reason, aluminum alloy is used for all pressure structural components.



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fuselage design improved with

MAGNESIUM



Thin skin magnesium air frame construction offers many advantages.

The illustration at left shows how magnesium skin has simplified construction of this fuselage section. Note the absence of longitudinal stiffeners which not only makes production faster and easier but gives more interior space.

Other advantages of magnesium sheet and plate are greater torsional rigidity, high resistance to impact, less critical corrosion and better contours because thin skin permits flush riveting.

To meet critical problems of weight and rigidity, look to magnesium, the world's lightest structural metal.

THE DOW CHEMICAL COMPANY

Magnesium Department • Midland, Michigan

New York • Boston • Philadelphia • Atlanta • Cleveland • Detroit
Chicago • St. Louis • Phoenix • San Francisco • Los Angeles • Seattle
New Orleans • Toronto, Ontario, Canada

Dow



with explosive decompression—one while on the ground during a test program, another while in the air—have indicated the accuracy for elimination of progressive type failures.

• In the failure during the test program, a pressure burst on a flat-type pressure bulkhead had sufficient static loading of the compression flange. This resulted in a complete failure in bending of the bulkhead splitters and web and the entire bulkhead was destroyed.

• In the second section, a hemispherical oil type bulkhead in the air cabin blew out in flight while slightly under maximum precompression loads. Analysis of the failure, Datta says, indicated that stress change had possibly occurred in the bulkhead web, which resulted in a complete loss of the bulkhead. In this case, the bulkhead was operating at a stress level of approximately 35,000 psi.

While longitudinal bulkheads carrying pressure loads in hoop tension are considerably lighter, they are not particularly adaptable for shear reinforcement such as is required in the B-35 where, due to the hood box, continuity of the monocoque structure is not possible.

This also presents problems where space is at a premium, therefore, can not be used in all applications. It is believed, however, that hemispherical type construction can be made safer from the standpoint of progressive type failures if the bulkhead web is strengthened at a sufficiently low stress level.

Datta reports that the new members who occupied the cabin at the time the progressive failure occurred, even though one member was seated within three feet of the fire of the bulkhead—the rate of decompression was somewhat cushioned by expansion of the air into the inflator eyes and also, the opening was so large that the velocity of escaping air was relatively low.

• Small Balsa Analysis—Another design is present in major failures of the type described. That is the possibility of damage to pressure structure or vital installations by failure of the overall system from flying debris and expansion of air into areas which are not capable of taking even low internal pressures.

Notions for adequate structural protection preventing this have been discussed in the test program that have been run on the B-35. The experience gained from these tests has shown that the primary problems in producing an adequate structural design are not basic structural considerations, but are sound design of the detail design which can escape analysis.

The production test procedures on the B-35, Datta relates, are to test each airplane to the maximum normal operating pressure and periodically to check individual airplanes to a pressure load just under the design limit pressure.

The first production airplane is checked for significant change in static weight which affects the structural integrity is also checked at these higher pressures.

Compliance toward the pressure relief design is difficult to avoid as long as no trouble is experienced, and the maximum design energy is not processed or is not fully realized until the effects are released.

• Transparent Sections—Datta holds that most important by far, from the usual point of possible failure, are those areas of a secondary structural nature, such as windows and doors.

The B-35 uses horizontal plate glass throughout the fuselage, and laminated plastic for the opening between the dorsal plate and the window structure. The glass used in the endowings is a combination of the expanded metal-mesh type construction. This design has proven very adaptable from the first design loading support for the window glass on the endowings structure. There is, therefore, no spring action. This is considered essential if sea level pressure is used, and the enclosure glass can get very strong from solar radiation.

A considerable decrease in glass strength occurs with increase in temperature at the plurality of the steel-to-metal interfaces and the glass becomes very strong, secure loads in such elements rather than in a homogeneous material.

Unfortunately, stepped-edge glass does present some installation problems, because clear tolerances between the glass and the structure is required. Probably the most undesirable characteristic of glass is the weak transition in strength, which requires high margins of safety and a consequent weight penalty. Another difficulty is external safety sensitivity, particularly in transparent glass, which makes inspection extremely difficult, because a small scratch

coefficients of expansion of the metal framework and the glass.

While there have been no difficulties of a serious nature with this method of installing the glass, there have been several problems involved. The main difficulty is that, due to the metal expansion causing the entire pressure loads from the glass into the supporting framework, a gross failure is set up which tends toward deformation. This has been a source of trouble primarily in the glass joints which are sealed for airtightness.

• Deformation Data—No longer, however, have occurred directly as a result of deformation, however, the investigation of a glass failure which occurred in a pressure flight declined a problem which had not been anticipated. It was found, after extensive laboratory tests, that a failure can occur in a point which is deformed and is subsequently subjected to cold temperatures.

This type failure is attributed to the contraction of the metal joints at a rate approximately five times that of the glass, which sets up shearing forces between the vinyl and the glass. These forces are of sufficient magnitude, if the deformation is extensive enough, to fracture the surface of the glass. If the glass is of semi-temper, this then destroys the equilibrium of the glass plate and results in the failure of an entire panel.

To prevent deformation, it was found that external restraining means were required on these glass panels which were subjected to heat along with pressure.

An improvement over the extended metal-to-metal contact type construction, Datta says, is an extended metal-mesh metal stepped-edge design. The steel glass lamination in this type extends past the edge of the other lamination to give them bearing support for the inner glass on the endowings structure. There is, therefore, no spring action. This is considered essential if sea level pressure is used, and the enclosure glass can get very strong from solar radiation.

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SHOWED-UP SCORPION

This Northrop F-86C Scorpion gets around during cold weather without maintenance tests conducted by the Air Force at Luke AFB, Phoenix. A-1000 test aircraft was checked and operation and maintenance tests were made. Radar sections flown by

the cold proved out equipment under the extremes of cold, and as in low power trials were completed in relatively short periods by Altona Subgroup with its own. In 500 mph runs it is operated with light-maintenance squadrons.

Aeroprop-

Ruggedness Starts

with...



Design



Building for today
Designing for tomorrow

Aeroproducts

AEROPRODUCTS DIVISION • GENERAL MOTORS CORPORATION • DAYTON, OHIO

Ruggedness comes straight off a drawing board, as the engineers at Aeroproducts will tell you.

For an idea, drawn in all its details, was the beginning of a great new Aeroprop—the first propeller to successfully handle the enormous power of turbo-prop engines. Yes, from this drawing, came the ruggedness of the dual-section Aeroprop for planes of near-sonic speed.

But the design produced was a more than ruggedness. It produced the reliability and service control which resulted in having the Aeroprop specified for the U. S. Navy's XP5V, B3V, A2D, and the A2J.

This great turbine propeller is reversible—can handle nose takeoff and smoothly, electronic governing and synchronizing circuits control turbine speeds automatically. And Aeroprop's self-contained hydraulic system makes installation and maintenance a simple matter.

Aeroproducts engineers who are among America's foremost propeller experts—are available to you for consultation on any propeller application in the subsonic, transonic, or supersonic ranges. Your inquiries will receive prompt reaction.

or such an engine strength actually.

► **Plastic Quality**—The plastic righting blades are of laminated construction and consist of outer laminations of methyl methacrylate with an inner lamination of polyvinyl butyral. The mounting consists of rubber extrusions bonded to the base of the drive which, in turn, are clamped around the structure.

The laminated plastic material, Data says, has many excellent qualities, among the most important being good optical properties, light weight, resistance to corrosion, and resistance to impact.

The undesirable factors are its poor abrasion resistance and rather low strength. The material's poor abrasion resistance and its reaction to some chemicals used in aircraft maintenance present a problem.

Even in view of its objectionable factors, the plastic material has proven excellent in service from a performance standpoint and there have been no known instances of a blade being lost, he reports.

No service difficulties have been experienced with the optical plastic glass center windows. There have been no known instances of a blade being lost, he reports.

► **Door Practice**—All doors, both external and internal, are designed to open inwardly. This is contrary to current design trends, but it is left that inverted-opening doors, in which pressure can be raised around the mirror framework of the opening, offer sufficient advantage over outward-opening doors, where pressure loads are caused at a few concentrated points, that consideration should be given to this type design, Data says.

A compromise solution to the door problem might possibly be the use of inward-opening doors on those sections which are frequently used and outward-opening doors for emergency escape in which the locking mechanism can be adequately selected to prevent inadvertent unlatching.

An associated problem that occurred on the B-35 doors was the providing of adequate door seals that would retain door flexibility and not seal, at low temperatures. It was found that the usual rubber compounds were not adequate and silicone-type rubbers were used to eliminate leakage.

B-35 experience has not indicated that there are any unusual problems in sealing of structural units for high altitude operation. Flexible sealable commercial sealing compounds are adequate for the temperatures and pressures encountered. Cabin sealing is accomplished by Thiolite-type liquid and is throughout in the B-35.

► **Heating**—There have been no major operational problems in the present



ALUMINUM CYLINDER holds radio-controlled life raft before launching.

Rubber Raft Can Cruise 300 Mi.

A radio-controlled life raft developed by Douglas Aircraft Co.'s Long Beach division, in cooperation with the Air Medical Command, starts out as a 38 in.-long by 21 in.-diameter aluminum cylinder and ends up as a 28 ft.-long by 8 ft.-wide rubber boat.

The transformation begins to take place within two minutes after launching, when the cylinder begins to open under pressure of carbon dioxide. Gas forced expansion inflates the cylinder-based rubber skin to its full shape. Shown below suspended from the raised gantry is a rubber raft.

It is a perfectly identical to the original installation. Ejector problems have been providing adequate heat under extreme low temperature conditions and proper distribution throughout the cabin.

The heat distribution problem is complicated in the B-35, says Data, because of the large size of the cabin and the fact that the crew stations are at various levels within these compartments. Cabin heating and pressurization air is used for the engine and blower defrosting, and this heats the temperature to a value which will not cause overheating.

To solve the problem of providing adequate heat and to ensure the distribution, General-designed electric heaters are used at these locations where spot heating is required.

Effects of solar radiation also may create the heat distribution problem. The large bubble pilot's canopy, the highest point in the fuselage cabin, is provided with a separate recirculating duct system which takes cooler air from a lower compartment level.

Naturally the adequate defrosting provisions of transparent areas was established during the flight test phases of the airplane. Deicing-type de-icing panels were used and did not

disturb in climb alone.

Equipment includes a four-cylinder radial engine and fuel for 100 mi. of cruising. There's a remote control radio system, two-way radio communication set, landing system, automatic pilot and boost and control gear to carry eight occupants for a five-day stretch.

The place for vessel launching the raft will, by radio control, be able to start or stop the engine, steer the boat to maneuver or set it on a desired course course.

First tests recently proved the unit's operation phases.

prove entirely satisfactory. The test aircraft employs hot air defrosting of all transparent areas in various stages.

► **An Source-Data** points out that, while flight at high altitudes has presented a number of new problems and suggested errors, there are some fine points of the present system which will be simplified in future aircraft designed for high-altitude operation. An example of this is the availability of a cabin air source such as the turbocharging air on the B-35, thus eliminating separate mechanical-driven compression.

Future aircraft which employ hot-bone-type powerplants will use air to provide a source of pressurized air, but will also provide heat eliminating the necessity for separate heaters. A high pressure air source, such as is available from turbine bleed air, also permits utilization of the air cycle expansion turbine for low-altitude cabin cooling.

An example of this type system was required in a special oxygen installation in the reconnaissance airplane, where precise temperature control was required. Air was taken from the last compression stage of the J47 turboprop and ducted to two 15-cu ft expansion turbines. A modulating bypass valve was provided around the expansion turbines to permit raising of the last compression

Valve Talk

for WM. R. WHITTAKER CO., Ltd.

By Marvin Miles,
Senior Manager, Aviation Military Apps.



There are torture chambers at the Wm. R. Whittaker Co., Ltd.—strange rigs of coils and tubes and dials, of flumes and dry ice, of rushing air blasts and gurgling hydraulic pressures.

They are a most important part of the Southern California valve centers, for they test the performance of every new valve design under a variety of simulated operating conditions.

No valve is better than its performance, regardless of its size or cost. To meet perfection, Whittaker tests its valves to limits far beyond those set by customers. They're tested to exhaustion, to destruction.

Though few know the search in which these valves function must be ready for stress with a minimum of delay, whether they be in duct, pipe or Arctic zone. With the space of a few hours, a valve may meet rugged conditions of loading tests, extreme cold, low and high humidity, corrosive atmospheres, rain, snow, hail, dust, ice, sand, dirt and fog.

And if its valves don't perform, a work performs.

"At present, Whittaker test labs are running qualification tests on 40 different units," says E. C. Glick, Vice President in charge of Engineering. "These include a half dozen different environmental tests. In addition, we're putting 50 development valves through the torture chambers."

Thirty-five Whittaker test rooms and equipment systems, \$150,000 worth of test equipment in nine laboratories, enable many of the special procedures and techniques required, rig up the toughest operations, follow the tests, translate the results and write the reports.

Half of these extensive lab facilities were completed within the last year to meet increasing demands. Even so, Whittaker's test area cannot stay abreast of the work load. They'll add another \$1 million in needed work is assigned to approved commercial test labs.

"Testing is a tricky business," explained Frank McCard, Chief Test Engineer. "Design development testing must precede the standard data. Back information is accumulated and used."

"When you select the flange, make it a single and to a wide range of temperatures, you can predict the results of different coefficients of expansion and friction?" Who knows

when test stress will occur, how great variations will change, what the test part will be?

We said destruction three factors usually. With some stresses repeated constantly keeping new problems, we must continually keep up on technical progress.

"Temperatures in the systems studies range from minus 85 degrees Fahrenheit to plus 2000 degrees, the cold induced by dry ice and alcohol, the heat by gas and electric heaters. One test rig could test, involving a few of 50 psi test per minute, 500 PSI pressure, at minus 65 degrees, required nine tons of dry ice."

Through use of power-offices, variable-flow pumps, Whittaker achieves the extreme hydrostatic and low pressure encountered in jet flight, and the valves check and wire through proving life-cycle tests with maximum regularity.

The search is ever increasing. In one section of the lab, a device and test electrical gauge is releasing a unit from 10 to 30 million times at varying speeds of from two to thousands of cycles a second. In another section, data study the conducted and radiated wave noise in a stable ratio to noise and radio interference.

In yet another test, used or done in varied across a valve in 1200 test specimens for 24 hours to test the valve's better efficiency in a fluid weight device test.

Another valve may be working it out for 15 days in a laboratory chamber to test for corrosion and maintenance. Still others checked with air and fluid in a high pressure chamber, are watched as they function in atmospheric conditions. And all the while, intricate inner and metal element and hydrodynamic of these complex developed are following the test for complete details of performance.

It's hard to believe that valves can stand up under the extreme conditions of the test lab—but they do at Whittaker's test center.

as from the supply and the cooled out from the turbine discharge. Duct was 2-in. aluminum tubing, considerably easier to fabricate and install than the larger tubing necessary in the low pressure system.

Turbine systems, Dietz says, will present new problems, among them being the need for providing ducting material, such as titanium and stainless steel, capable of taking high-pressure and high-temperature air, and the necessity of providing for fast thermal expansion.

Pressure Demand Oxygen Current production 8-lb flow D-I oxygen regulator. There are pressure demand type and self supply 200% oxygen at 10 mm Hg at 10,000 ft. While the equipment is effective in accidental decompression at altitude, considerable modification is required in its use, says Dietz. Pressure breathing requires considerable physical effort and can be continued only for about 10 min. The length of time, however, is sufficient to allow a pilot to take altitude where the manual demand oxygen system can be used.

Adaptivity of the pressure demand system was proven in an accidental decompression during a test flight above 40,000 ft. on a B-56. While the unit was repaired, the need for rapid training and operational procedures was brought out.

Present standard operational procedures require that the oxygen supply must be in a ready position when at an altitude of 20,000 ft. and be on oxygen above 40,000 ft.—45.

Stress Rule Aids High-Temp Research

A new approach for determining high temperature material data has been worked out by two General Electric Co. metallurgical research engineers. The key is a stress formula which is built through with time and temperature. Report is that application of the formula, which has been embedded in collector form similar to a slide rule, can save as much as a year of test time.

GE says that with data obtained on short-time creep tests and the otherwise, long-time strength of metals under high temperatures may be obtained.

Other advantages reported include getting of data at intermediate temperatures without cross plotting, and approximation by a single curve of complete creep characteristics of an alloy to facilitate comparison with other materials.

The formula was developed by James Miller and Frank S. Larsen of GE's Thompson Laboratory.



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Photos Give Clues to B-52

By David A. Anderson

One of the more unconventional systems on Boeing's big B-52 bomber modifies lateral control of the craft. As yet Boeing and the Air Force are not talking about how the airplane is rolled or banked. But there have been enough photos of the plane in various attitudes to give some indication of how the job is done.

From those photographs comes this speculation about the lateral control system of the B-52.

There apparently are two separate sets of control surfaces which could affect the lateral motion of the big bomber.

• **Section One** looks like a spoiler. Mounted on the upper surface of the wing in the region of the outboard engine nacelle, it is made of three sections.

It is possible each section operates independently. An upward deflection of about 60 degrees is shown in one picture.

• **Section Two** resembles no other conventional surface. It is located just outboard of the outboard nacelle at the wing trailing edge. It has a short span, and fits between the two segments of flap. Deflections shown in the pictures are moderate and differential, like aileron deflections.

• **What They Are**—So much for what the surfaces look like. Now let us try to imagine Boeing's reasons for this unusual layout.

First of all, as the B-52 the wing is extremely flexible. The airplane is a high-wing type. Those two factors spell complications in the use of conventional ailerons at the wingtip. Aileron loss or reversal are only two of the observed effects on lightly swept wings operating at high speed.

The physical dimensions of the deflector make a tough job of transmitting control motions from cockpit to ailerons at the wingtip, adding a third complica-

tion to the list above.

So all things considered, it's reasonable to believe that the B-52 control system works like this:

• **Section One**. This is apparently nothing more than the spoiler it looks like. The only picture that shows it deflected has the B-52 banking on the ground. All-Flight dials so far have been taken from below the airplane and have not shown the upper surface of the wing at all.

• **Section Two** is used fully or in lower. This evidence is reached because of the differential deflection of the surfaces shown in a picture of the B-52 in a banked attitude. There is also a trace in some test clearly visible on the trailing edge of the starboard aileron, and spoiler or flap do not have tabs.

Furthermore, the aerogenic location of these surfaces puts them where the wing is relatively free from both twist and bending, which simplifies aerodynamic and mechanical problems.

Putting all this together, it seems as if the B-52 has a lateral control system composed of spoilers and ailerons. The spoilers are on the upper surface of the wing located at about the midspan point.

Ailerons are also positioned at the midspan point.

• **Aileron System**—Reason for this unusual system is also a matter of speculation, but it does seem possible that the ailerons alone would not get enough rolling moment for cruise action in example. They probably would suffer for steady flight or long stretches on autopilot. In the event of a light, the B-52 spoilers would probably add their forces to the system.

That's what it looks like from the photographs. As with any other preliminary based on incomplete knowledge, these paragraphs must be overdone. But in the writer's opinion, the speculation seems logical and possible.

Lateral Control System



AILERONS of the B-52 show in this light shot at the post booster. Note differential deflection of the surfaces, the short span and the unconventional location near the mid-span point, probably chosen because of large wing deflection at tip.



SPOLERS bend the down lines of the Boeing B-52's swept wing. These surfaces, made up of three sections probably capable of independent operation, form part of the lateral control system. Spoilers reflect adverse yaw characteristics with flexible wing.



LIGHT SHOWS THROUGH the wingtip of the Boeing B-52. Complexities of structure and linkage and aerodynamic problems have dictated this surface as an aileron and forced its use in a later slow banked maneuver.



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How Saab Builds the J-29 Flying Barrel

The J-29 highlights Sweden's capacity in advanced aircraft design and production. This jet fighter, fabricated by Saab (Svenska Aeroplan AB) in a modern production facility at Linköping, is considered a first-rate plane.

Though dubbed the Flying Barrel because of its paraboloid body, the J-29 is a clean and compact configuration. It is a "lean-grow" product that typifies the country's industrial independence—even the plane's de Havilland Ghost engine is built in Sweden, by Svenska Flygmotor AB.

► **Outside-In**—The plane incorporates thick sheet, heavy sections and computered shaping. To maintain accurate contour on components, Saab employs a laydown scheme working from outside in. The outer skin surface is controlled by exterior pickup points in an assembly jig and the supporting struc-

ture is built up to fit the skin's inside contour.

England's Fairey Aviation Co. used a similar production method for its Puma ST (Aviation Week Jan. 18, 1980, p. 27).

Reverting on small subassembly components is accomplished at work stations along either side of a conveyor belt operated from an inspection station at the end of the line. Completed subassemblies go to a production store until ready for the next operation.

Each large stationary assembly jig carries a serial number and is painted in an individual color. The number helps the planning department follow the work and the color is used to avoid moving location fittings from different jigs. Large boards are positioned along-side each jig, with places to hang each fitting.

After the three fuselage sections—nose, center and rear—are brought to-

gether in the mating jig, the completed assembly is put on a trolley. This carrier is then moved until it is ready for flight test.

► **Plane Data**—Production J-29s are 41 ft 10 in. long, with a 5,600-lb thrust Ghost engine, and are designed for a top speed of about 650 mph, although this speed has been exceeded.

► **Design** accommodates such equipment as landing gear, fuel tanks and armament. Length of the body is 33 ft 24 in., height is 12 ft 31 in. Wings are conventional two-spar structures with 25 deg sweep. Span is 35 ft 1 in. Wing leading edge is fitted with non-magnetic ducts to improve stall properties on landing and at low flying speeds. Stabilizer, mounted high on the tail is electrically adjustable in flight.

Four sections of the J-29 have no holes drilled to the wing behind the rear spar, but later models are designed for fuelage-controlled brakes.

SAAB-29 CUTAWAY



MATING

Nose, center and tail are put together here. Four legs then hold trolley for remainder of line.



INSPECTION

J-29 gets thorough inspection here before proceeding down the line.



ENGINES

Hydraulic bulk de Havilland Ghosts get final touches on subassembly line before going to planes.



END OF LINE

Planes now flight test after grilling wings, stabilizer and landing gear.



BEFORE: Operator had to leave known rollers with handwheels, then crank pedal to position for new rivet row.



AFTER: With new remote control lateral shifting (foreground), operator steps out. Result: faster production.

Mechanical Table Speeds Temco Riveting

Riveting speed is being pushed up at Temco Aircraft Corp., Dallas, Tex. The company has fitted a wing support table, used in conjunction with the first General Dynamics Riveter, with a new push-button-controlled lateral shifter that is saving it 30 to 40 sec. on each RV-5 wing panel it is fabricating for Lockheed Aircraft Corp.

Before and After: The Lockheed-designed table which was furnished to Temco for the new job had provision for remote control longitudinal shifting for driving a row of rivets, but lateral shifting was a manual operation. (Aviation Week Dec. 24, 1955, p. 36). The operator, after operating a line of rivets was completed, to go first to one end of the table then to the other in order to loosen clamps, shift the panel with a crank, and again tighten the

clamps. This was time consuming and fatiguing during the day's run.

With the new Temco lateral shifter, the operator doesn't have to leave his seat to move the panel. This means faster operation and less fatigue. Other advantages the company expects include greater accuracy and precision of placing rivet rows. The Dynamics also drills out and replaces lock rivets.

Components: In addition to the lateral shifting mechanism, the new installation includes a remote mechanism (driven from the control line) to which panels are bolted and actuators operate after the shift, a solenoid clamp to maintain tension on the retaining bar except during lateral shift; a group of electrical and mechanical safety provisions to prevent damage to table, clamping or panel or to rivet or

operator nose, and the actual control mechanism.

Simple Changes: The installation is mounted directly on the Lockheed table and involved only minor modification—groutwork was being in the carriage where rollers and solenoid clamps replace handwheels and mechanical clamps. Only one addition is a 66-in. length of trolley track carrying the remote-control cable on rollers.

Total cost of the lateral shifting installation was just over \$3,500 for material, labor and purchased parts. Cost of work fix surface table modifications scheduled on a second Dynamics will be considerably less than that figure, Temco says.

Complete takeover on the first job was accomplished with only six hours of machine down time.

USAF CONTRACTS

Following is a list of recent USAF contracts announced by Air Material Command:

- **Avco Instrument Co.**, 1205 Madison Ave., Cleveland 16, Ohio, \$700,000 for 100,000 units.
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New Marine Jack-of-all-aircraft—Latest addition to the helicopter fleet of the U. S. Marine Corps is the Sikorsky HO4S, now being delivered in substantial quantities.

This is the third type of Sikorsky helicopter to be used by the Marine Corps which has pioneered many revolutionary combat tactics with helicopters in actual combat in Korea.

This type, also in service with the U. S. Army Field Force, is a four-place development of the earlier Sikorsky HO4C, holder of the world's speed and altitude records.

In service with the Marine Corps, the new HO4S helicopter is expected to be of great value as an observation-liaison aircraft and for evacuation of wounded and trapped men.

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Old Voltage Problem Gets New Answer

- Navy buys GE magnetic amplifier regulators.
- Will install the units in its new aircraft.

By Philip Klum

Schlesinger-Navy's Baker is trying to eliminate lightning to solve one of the oldest problems in aviation—that of providing a rugged, reliable, long-lived device to maintain constant voltage output from aircraft generators.

The new voltage regulator, developed by General Electric, uses static magnetic amplifiers to replace the aging, limited number of previous aircraft regulators. It is designed to regulate AC generators (alternators).

Baker has ordered the new GE regulator for use on many of its new aircraft, including the McDonnell F-1H, Chance Vought F7U-3, Douglas A-1H, and North American A-1H. The Air Force is also reported to order the GE regulator.

• **GE Claims**—Based on experimental flight tests in a B-36, and extensive ground tests, GE says the device will:

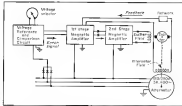
- **Hold constant voltage** within $\pm 1\%$ over a load to full load.
- **Hold voltage constant** to $\pm 1\%$ between temperatures of -60°F and 160°F (at constant load).
- **Return voltage** to 95% of normal voltage less than 0.1 second after operation or removal of full load, full fluctuation up to approximately 60 kva rating.

• **Be unaffected** by acceleration of 10G in any direction or by severe vibration.

• **Be long-lived**, with expected trouble-free life in excess of 5,000 hours.

The only drawback is that magnetic amplifiers need to operate from a power source which provides their own regulation to the power sources. However, most new military aircraft are going to be power sources.

• **No Moving Parts**—GE's regulator has no moving parts, which one counts a rugged advantage when operators only need to change a constant in parallel circuit, building up alternator voltage. This, coupled with its other features, explains why Baker is willing to accept a heavier 17-lb regulator to replace the 12-lb unit previously used.



MAGNETIC AMPLIFIERS take place of moving structure in GE voltage regulator

► **Troubled Pair**—The existing aircraft which has been the heart of many voltage regulators was also the root of their troubles.

In previous aircraft voltage regulators, generator output voltage has been used to develop a magnetic force which works against a spring to position the regulator armature proportional to the generator voltage. "If there's high voltage," the armature was designed to have very small travel from a condition of low voltage to one of high voltage. As a result, high G maneuvering loads or phase variations could upset

severe electromagnetic coupling failure to give erratic generator output voltage.

Until 1944 the regulator armature needed a mass of iron fingers which acted as part of movement of the armature in the generator field circuit to vary field current. In service, the fingers suffered from "bakeburns" at the levels of unpermitted personnel. As a result, the fingers became pitted, bent, and deformed.

► **Better, But** —In late 1944, the much-improved coil-type type regulator had begun to replace the finger type. Here the armature mechanism controlled the pressure on a stack of carbon disks to change their overall resistance which in turn varied generator field current.

Frage adjustment and getting fresh fingers were eliminated, but aging and arcing of the carbon disks took their place. With its moving armature, the coil-type regulator was also susceptible to severe aircraft acceleration vibration. Whenever mounts were used to further reduce this effect.

When Baker let contracts for the development of a line of aircraft alternators several years ago, the contractors also called for a "bancable" new and improved voltage regulator. The new GE regulator was the result.

► **Operation**—The regulator is designed to operate with a three-phase, 120/208V, delta or wye-connected alternator. The alternator voltage in all three phases are rectified and combined to give a single dc voltage proportional to the average of the three voltages which feeds the regulator. This

dc voltage is compared with a constant voltage reference in a bridge circuit which develops an "error signal" whenever the alternator voltage varies from selected value (See diagram.)

The resulting voltage reference is provided in a JAN-OR2 solid cathode tube. It is a considerable source output than a conventional vacuum tube. Being so different, grid, or heated cathode. However, any component which uses a single rugged voltage reference in a glass envelope is frequently "suspect."

GE has added a "backing" in the form of a second JAN-OR2 tube which is connected in parallel with the first. As long as the first tube is functioning, the second serves as a "standby." However, it will instantly take over the voltage reference should anything happen to the first tube according to H. W. Garck of GE's Industry Case Department.

► **Push-Pull Stage**—The first magnetic amplifier stage consists of two small adjustable reactors, connected push-pull. Each has its control winding (physically separated from a common field grid) connected in series opposition to the voltage compensator field. Each winding controls the current flowing through the reactor's output winding from the alternator through the reactor's output winding.

When alternator output voltage is at the selected value, there is no error signal, and balanced output will flow from the two first-stage amplifiers.

This first-stage output is rectified and applied to the control windings of the second-stage adjustable reactor. The output winding of one of the first-stage amplifiers is connected to the "back" control winding in the second stage, the other is connected to the "load" control winding. All alternator voltage is "on the bottom" equal currents will flow from the first stage through the "load" and "back" windings of the second stage.

If alternator voltage varies from the selected value, the error signal applied to the first stage will cause it to increase the flow of current in the appropriate "back" or "load" second-stage control winding. This in turn increases or decreases the output current from the second stage, which in turn rectified and applied to the input field of the reactor.

► **The Reactor**—The reactor is a small dc generator which is built into the alternator and driven from the same shaft. It provides the adjustable high-power level of resistance for the main field of the alternator. In a sense, the reactor is a constant stage of amplification to raise the power level of the output from the voltage regulator.

The polarity of the error signal from the bridge compensator is determined by whether the alternator voltage has

more than or below the selected value. This signal determines whether the push-pull stage passes the proper amount through the "load" or the "back" windings of the second stage. This in turn determines whether the reactor field current is increased or decreased, thus controlling alternator field excitation and hence its output voltage.

► **Feedback**—The voltage regulator is a "closed-loop" system in the sense that a regulator change in reactor field current causes a change in alternator voltage which will in turn correct the initial error signal calling for the change in field current. However, within the reactor loop, GE has included a feed-back circuit to increase system stability and limit voltage overshoot.

The compensator system GE is developed shows details of the feedback circuit except to note that it provides the regulator to "unstrap" the second stage always in alternator output voltage.

► **Iron Windings**—The alternator circuit has an extra field winding (not shown in detail, diagram) which functions as a heating device. It is excited from rectified alternator output and acts to partially "back out" the main reactor field. This allows the second-stage reactor to supply sufficiently high line voltage to operate on the proper portion of its saturation curve.

Because of the special center field required, the GE reactor regulator



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cannot be easily substituted for existing electronic regulators in A-C systems. However, the regulator can be modified to provide reasonably satisfactory operation on most alternators whose exciter line is in the field.

The experimental B-30 flight tests were conducted using Westinghouse alternators without bus fields. The results were very satisfactory, according to E. S. Gallagher, sales manager at GE's Aviation Division.

• **Field of the Problem**—Evidence of the new regulator's rapid response and stability was shown to the writer by K. W. Carlson of the Aviation Division on a test setup which included an exciter-drive alternator. Carlson placed a direct short across one alternator phase and then removed it. The voltage recovered to constant about 25% then returned to normal in slightly less than 1 second.

Carlson attributed the rapid overshoot to the alternator's high inductance reactance. If its purpose designed to be high so that direct current cannot be limited to about three times normal full-load current in order to "burn out" a short circuit. The exciter built for only about 0.35 seconds, so a worst design is also that component in the line.

A lock made the CIP regulator drives that start of the spin as the unit is activated by selection switches rather than by the magnetic amplifier themselves. If and when the present 60°C temperature limit of germanium switches can be used to permit them to replace the selenium rectifiers the size and weight of the regulator could be reduced considerably.

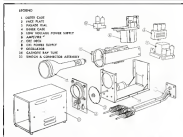
Because of the absence of moving parts and the ruggedness of its components, the regulator requires as much as vibration mounts.

• **Speed Range**—Actual alternator can be divided into two speed types. One is driven through a constant speed drive which operates alternator frequency between 320 and 450 cps. The other is directly driven from the engine which is alternator frequency range of 400 to 500 cps.

The response of magnetic amplifiers is affected by the frequency of their power source, in the case of the magnetic amplifier. Consequently CIP found it necessary to come up with two slightly different regulators, one for each type of alternator. The basic difference is the addition of a special frequency-compensating network at the wide-speed range regulator.

Ultimately, Gallagher says, it may be possible to design a single regulator capable of handling both types of alternators. Better regulator is presently capable of handling alternators of different line ratings in its own particular speed range.

EQUIPMENT



LAND-AIR analyzer weighs 22 lb., costs \$1,625 plus accessories and taxes.

New Analyzer Interests Carriers

Bendix conference discusses Land-Air's lightweight unit; personnel training problems also evaluated.

By George L. Christian

Detroit, N. Y.—Latest entry in the electronic engine analyzer market, the Land-Air, aroused considerable interest at the recent Bendix-Servitor sponsored Dayton and Analytic Conference here. The Land-Air may be used as an airborne or a portable engine test.

Highlights of conference presentation and the three basic types of engine—portable, airborne and portable—were reported in this section July 28.

The Land-Air instrument was developed by Loren Serfaty, who is chief engineer of Land-Air, Inc., in Chicago, Ill., was associated with the Sperry analysis.

The Land-Air analyzer is a unit by the Navy, has been evaluated by American Airlines, and is being tested by USAF, according to Ben Postulak, assistant to the president of Gulfstream Eastern Airlines. (Land-Air is a subsidiary of GEIA).

• **Land-Air Analysis**—Paul Keville, AA engineer, reported his airline's experience with the Land-Air analyzer. His evaluation was limited, he said, because the analyzer was late and not preferred to point of acceptance when tested, nor was it being used in production quantities. But he did single out three

good points of the Land-Air:

- Weight is a 22 lb.
- Size is 7 1/2 x 11 x 15 in.
- Simplicity is outstanding. The analyzer has only one dial to turn. The main control has six tabs (half the number of other analyzers) and the wiring is simple.
- Sensitization of instrument to engine is through a modified standard AN instrument generator.
- Determination of engine test are looking at can be done quickly and accurately.
- Dynamic testing of an engine is possible with the instrument.
- Vibration analysis, as well as engine analysis is possible with the instrument.
- Some analyzers may be portable—airborne or airborne.

Keville concluded that the instrument was worth considering, at put was not ready in time for American.

- **How it Works**—Three patterns of one or both engines may be viewed as the B-A analyzer will simultaneously be analyzing the large knob under the scope (parade dial) each cylinder's pattern may quickly be viewed. An pattern may be left on the scope as long as required for interpretation.
- Vibration patterns are available to

install on one or more cylinders. The data used are the magnetic structure type and current vibration via voltage signals.

The analyzer may be broken down into eight subassemblies which can be quickly taken down to speed servicing and maintenance. The instrument will operate on 95-125 v. a.c. at 50-1,600 cycles or on 60-400 cycles.

High-inertia carbide may take makes patterns clearly visible without shock at all times, according to the manufacturer.

Postulak said that the analyzer had gone into production at a Chicago plant. He quoted the base price at \$1,625. Accessories may add up to another \$500 plus labor for installation. Servitor and \$5,380 was the total cost of analysis including complete personnel of equipment for Continental personnel, when evaluated.

• **Training for Airlines**—Training, of personnel in the use of analyzer was discussed in some detail at the conference. R. L. Beane of Bendix Aviation Corp. stated that the analyzer is easy to use and all classes of maintenance personnel can use it if properly trained. Training in a number of ways is being held, he said, usually being a matter of a few days.

An increasing approach to introducing analyzers to a group of mechanics was expressed by K. W. Farnas, tech. asst. assistant to the director of engineering of TCA.

The instrument should not be shoved down a mechanic's throat, he said. Mechanics need "advising" period. This means to allow them to watch experts operate analyzers, thus translate a mere all interest. Some they will feel that the instrument will do all the things (check) for it and all the things (check) on the indication on the scope, Farnas said. He stressed that he wanted the pilots of operation—useful data cannot be lost.

Airline diffused concerning type of instrument that should be analyzed. Northeast engine mechanics, while Continental have it over to electronics men. TCA agreed that it should be used by mechanics, writing it should not be used by electronic and maintenance, that would be too much of a heavy.

FAA, which probably has more experience with analyzers than any other commercial airline, talked of the need for training. The analyzer is easy to use and most classes of mechanics can use the instrument personally, but trainers should know something about engine in general and engine system in particular. FAA said. The service has established a two-day analyzer training course.

FAA mentioned that in its engine area, training seminars at the over-

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FINANCIAL

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Carpo carrier has put out a 'sweetened' convertible debenture to help pay for new DC-6As and spares.

The airline industry is watching with

interest a recent \$2,590,000 debenture financing arranged by the Flying Tiger East. The financing was necessary to pay for new equipment on order, financing plans at other carriers which need additional funds will undoubtedly be influenced by the cargo operator's response.

The new equipment of the Flying Tiger is about DC-6As, aircraft, estimated to cost \$6,990,575, 1960 when 20% of the entire issue is expected to be retired in this manner.

It is probable, as with all convertible debenture issues of this type, that assignment to help retirement into common stock will provide the necessary cash payments to effect retirements through the sinking fund or at maturity.

Flower Potting.—This financing, in a broader sense, reflects the transition made achieved by the Flying Tiger since its inception more than seven years ago as the National Skyway Freight Corp. The company obtained a better financial picture, its successful conclusion of its application for a Franchise Certificate of Public Convenience and Necessity from the Civil Aeronautics Board on Feb. 25, 1959.

This certificate, expires on Aug. 12, 1974, gave the company legal status as actively engaging in the transportation of property across authorized routes across the United States.

More pertinent to the company, however, have been the operations conducted through companies from CAA to its parent, authorized as cargo or freight in the United States. The same contract law that considers as business in underwriting, charter for passenger and cargo activities largely under contract for the industry establishment throughout the world.

SBIRY Business.—For the first full year of its operations ended June 30, 1958, the company's operating revenues totaled \$410,000. For the year ended June 30, 1959, its operating revenues were \$1,512,075, and for the year ended Mar. 31, 1952, \$86,283,559.

Of all of the operating revenues of the company for the year ended June 30, 1959, and for the year ended Mar. 31, 1952, revenues from the Pacific military airfield, the largest

source of revenue of the company (one-eighth) is \$7,492,464 and \$6,920,171, respectively.

Revels.—Directors and senior officers of the Pacific airmail have also had very large in the company's activities. For the year ended June 30, 1951, at the total \$15,582,000 gross revenues generated from all sources, a total of 77% was derived from activities reflecting this "Revels" operation. For the year ended Mar. 31, 1952, at the total \$16,252,000 in gross revenues generated, about 80% came from the same source.

Red to Black.—Reflecting the character of its special operations, the Flying Tiger converted its earlier deficit into profits. After a net loss of more than \$1.1 million for the two years ended June 30, 1948, the company showed its first profit, \$133,519 for the 1949 fiscal year. This net operating gain increased to \$400,490 for 1950 and \$1,024,993 for 1951, with \$349,377 for the year ended Mar. 31, 1952.

The Flying Tiger debenture prospectus notes that "Even though the Corporation's current business has growth potentialities, the investment and savings of the Corporation may be substantially reduced in and if future requirements demand."

Considerable support to the company's operations could stem from any success in its application before CAA for authorization to carry air mail, in which parcel post and air express, in which application promises to see into convertible operations from the stock sold earlier, now carrying the mail carrying function.

—Selig Abelson

Delta Issues Stock To Buy New Planes

Delta Air Lines has floated a stock issue of 100,000 shares to help handle its aircraft purchase commitments. At current market levels, it appears that the sale will gross about \$1,000,000, with \$225,000 going for underwriting discounts. This additional cash will bring Delta's outstanding common to 400,000 shares.

The financing is a straight stock flotation. It will help Delta purchase ten Convair 440s and four DC-7s. The Convair, together with spare parts, cost a total order of about \$6,300,000.

The deliveries are scheduled to start in October, 1952, and continue through July, 1954, at the rate of one a month. The DC-7 commitment together with related spare will total a capital outlay of about \$7,000,000. This equipment will be delivered at the spring of 1954.

AIR TRANSPORT

Brow-Wrinklers in Airworthiness Talks

PILOTS WANT:

better manual equipment
larger navigation lights
better weathered visibility

CAA WANTS:

standardizing tests

INDUSTRY WANTS:

don't mix or judge standardization larger endurance test for engines

sented a strong recommendation for regulations requiring standardizing tests on future civil transports, as a basis of safety experience in Britain, and U.S. military transports. Manufacturers industry, viewpoint is that present regulations do not prohibit use of standardizing tests by any aircraft operator, if he desires, and that engineering opinion on value of backward fitting tests is divided.

A recent IATA engineering committee recommendation approved the standard fitting tests as a requirement. More data on the subject should be accumulated before change in present regulations, it is contended.

• Dividing provisions. ALPA is taking the lead in recommendations for improved manual equipment for emergency use. It is recommending that the use of "standby" lights be allowed for the better tail fining of engine nacelles, thus permitting refs to be installed, less wing, or saving refs in compartments or feet and all sections of fuselage, as well as to be accessible from outside as well as inside the airplane.

Other recommendations are for emergency use cell lights in places with repeat-type switches, multiple tie-pts, services in all transport aircraft, and future and spring loaded external storage or refs so they can be categorized out and inboard, with a dual line attached to the service. Another industry spokesman believes the objective is sound, but expect similar opinions and ALPA to resist study on it, with manufacturers participating.

Adoption of certain provisions, however, such studies are completed in any case. (It is understood that the U.S. Civil Guard, working with the American Airlines, currently is running a test on inspection manual equipment in connection with ditching.)

• Navigation and anti-collision lights. ALPA also is asking for larger diameter (18 in.) navigation lights, with four tiers, the tail lights, and better visibility, and flasher anti-collision light at the top of the vertical fin.

Other participants in the engineering review had not indicated their position in the light proposed in advance of the meeting.

• Cabin questions. Some of the other technical transport questions scheduled for discussion in the session.

• Provision for fuel jettisoning equipment. as follows to maximum takeoff and landing weights.

• Clarifying present requirements for the establishment of a demonstrated maximum.

• Structural ground load requirements.

and various load factors required including forward, upward, downward and sideward factors.

• Proposal for adaptation of fire extinguishers in aircraft. none one of the planes manufactured after Jan. 1, 1973, whether it should be made inoperative, adaptation of fireproof nacelle door of aircraft engine, new type smoke detector, means of de-energizing electrical circuits in fire zones.

• Proposal to establish a separate category for transport type helicopters is not endorsed by industry spokesmen who contend that categories should be defined by operating limitations, rather than by aerodynamic standards, and that fixing of design parameters at this stage in helicopter development will have the effect of stifling further development progress.

• Finally, industry opposes a proposal to regulate quality control by the manufacturers, under a system of the CAA inspection system. Industry analysts see this as an attempt by CAA to force manufacturers to obtain production certificates, because CAA is not adequately staffed to conduct continuous type inspections in the plants at the present time.

Long on Plans, Short on Funds

(McGraw-Hill World News)

Melbourne-Australia has big plans to expand its airports to meet the surging volume of traffic, but getting the estimated \$15 million in funds to carry it all out is another story not quite so optimistic.

First stage of development of Mascot Airport (Sydney) alone will cost \$18 million and end of building up Essendon Airport (Melbourne) has not yet been completed. There there is a main area valued of \$1.25 million now for development of Avalon Airport near Sydney.

Second Australia is asking approximately \$5 million to improve Adelaide Airport and Brisbane's Eagle Farm field is said to need extensions which would run \$2 million. First stage of construction of a new terminal at Lindbergh in Toronto will cost \$1.5 million.

USAF Interested

(McGraw-Hill World News)

Boeing-The prototype, engine built landing gear designed and built by Boeing (Boeing) and demonstrated by test throughout Europe, has attracted attention of USAF. The Indian design has been ordered by Air Force officials to demonstrate the landing gear at Wright-Patterson AFB, Dayton, Ohio.

*Turbo-jet test plant for BRISTOL Aeroplane Company Limited, England



*SILENCING and thermal insulation system by MUFFELITE

The new BRISTOL turbo jet test plant is generally regarded as being in advance of any test plant for a similar purpose in this country or any other.

The MUFFELITE silencing and thermal insulation system, designed and supplied by this company, suppresses the noise generated by two engines on test simultaneously and reduces the temperature of the engine exhaust gases by several hundred degrees Fahrenheit.

The success of any such scheme depends essentially upon collaboration, from its initiation, with the Architect and the Test Plant Design Engineers.

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Search for the Safest Plane

- Industry-government conference explores the line between safety needs and business needs.
- Each group pushes its own plans and just about every shade of opinion is represented.

By Alexander McSorley

The old shakedown of whom proper safety regulation stops and interference with business begins and increases industrial bills started in Washington over proposed changes in new aircraft regulations.

This basic tug-of-war has dominated similar government safety conferences since the early 1950s. And it's not expected that this two-week battle, called by CAA, will be different.

Conferees at the third annual talks are engineers representing almost all the major aircraft and engine manufacturers and airlines, and technical staffs of the Aeronautics Administration, Air Transport Union, Air Line Pilot Association, CAA and CAB.

• Big theme: Most interesting and controversial subjects are how to:

- **Propose to increase the endurance test of aircraft engine** capacity from 170 to 200 to fit type certification. Proposal is sponsored by CAA, but opposed by members of the aircraft industry's powerplant engineers. Industry contention is that some extension of hours of testing at cruising power on the test stand are not as significant as understanding the nature of the test run, so that it will be run at varying speeds, some like military engine test requirements. Such a test at varying speeds would be more effective during the present 150 hr duration than an extension to 200 hr under present conditions, if it's not.

• Pilot equipment standardization. This is sponsored primarily by Air Line Pilot Association, which has requested standardization of cockpit equipment standardization of cockpit equipment to be sufficient, as to classification of function and layout, and in the adequate of wing and tail logging provisions.

CAA Technical Development Goals is expected to submit definite proposals at the review meetings, on the basis of studies it has made.

Aircraft industry spokesmen are opposing any more detailed regulations on this point than already exist. They note that several different groups are now actively tracking the problem of variable flight safety. Standards, maintenance, SAE and CAA. When better design data are available, the manufacturers will follow through with changes in design features, and better visibility, then say, under less limitations than at detailed regulations are presented.

• Cockpit standardization. The aircraft industry automatically that CAA revises its regulations (Aeronautics 40-2) to agree with SAE recommendations standards on location and activation of cockpit controls and instrument panel arrangement. SAE still is completing its aircraft panel recommendations, however. It appears likely that this recommendations will be followed more or less in advance of action from other conferences in cockpit.

• Standardizing tests. CAA has just

Japanese Progress On Airline Plans

Japanese interest in setting up an international airline network is making progress.

Last month a three-man mission, comprising Japan Air Lines, arrived in the U. S., reportedly armed with a \$10 million credit from the Japanese government to aid them in buying surplus equipment, including planes. The credit already has cover financing at CNA headquarters in Oklahoma City.

Japan Air Lines and its rival, Japan International World Airways, which recently signed a recent maintenance contract with California Eastern Airways (Aviation Week Feb. 2), p. 55) are locked in buying popular Japanese ship-jug airplanes. Both nations expect to begin operations using DC-10s, but hope to get more Japanese types later. They plan to fly through Hawaii to San Francisco and to San Paulo, Brazil. Japan Air Lines also would like to serve London.

A draft of a Japanese U. S. air agreement, paving the way for these routes, was concluded in Tokyo only recently, but it submitted for signature to the Japanese and U. S. governments. In return for military goods Japanese carriers, U. S. airlines will get improved rights.

PanAm Asks Bids On Big Hangar

Pan American World Airways is taking bids until Aug. 12 for construction of a \$600,000 administration, maintenance and storage building at Boston's Seattle International Airport, near Seattle, Wash. The hangar will be capable of housing two DC-10s or DC-10s in an Stratoliner.

PanAm presently uses Boeing Field for its Seattle operations, but is following United, Northwest and Western to Seattle-Tacoma. Boeing Field also will be left open to the airlines and feeder lines.

Aer Lingus' Profit Climbs Sharply

The Irish air carrier, Aer Lingus, rose up impressive financial gains for its first year ended May 31. Profit for 1966 was \$18,184,161, up from \$15,085 for the previous year. Gross revenue for the current fiscal year was \$4,553,077.

Aer Lingus flew 3,474,776 hrs. in an increase of nearly a half-million miles more than the previous year and transported a net load factor of 79%. The carrier's all peak night arrivals rose



ALLISON TURBOLINER that accidentally dropped passengers but is still, back in.



AND DRAWS A CROWD: Trouble in field and plane will take some test flights.

Turboliner Nears Service Tests

The first U. S. turbo-prop powered transport, the Allison Turboliner, is now reliable enough for service test flights, say Allison and Aerodynamics divisions of General Motors.

However, an incident that occurred last month on its maiden "test" flight—two coasters, right from California to Indianapolis illustrates problems.

In re-developed prototype design, deceptively sophisticated with a quiet thrust, the all-new propeller-driven turbo-prop aircraft at Ft. Peck But test pilot Vance had set the piston, turbine phase down on the piston, coasted, with no more damage, than bent-out broken and torn. (They were insensitive because of power failure.)

Case of the trouble. The designers, set to change the turbine loss the prop in case, all power failure, were adjusted too late.

tended to show gains, up some 4,380 passengers to a total of 15,297. Aer Lingus, overall total for the fiscal year was more than 200,000 passengers.

Carriage traffic increased considerably—33% more than the previous year in a total of 4,800 tons of all types of freight. In addition, Aer Lingus earned 1,193 tons of mail, up from the previous year's total.

In view of the expected continued traffic gains, the carrier has placed orders for four Vickers Viscount turboprop transports and four Bristol Widows in order to bolster its present equipment.

Solutions arrived in the spring test case, so no part can trigger the alarm, but the more severe negative force of a power failure will.

The Allison Turboliner prototype does not have a device installed to re-propel engine and propeller after decelerate, but future configurations will incorporate such a device, Allison indicates. After more development is necessary because in case of power failure a windmilling turbine propeller combination with the more thrust, could be fatal in takeoff.

Except for the deceleration accident the three stage flight from California to Indianapolis was "completely successful," Allison says.

New Turboliner is at West Coast Airport, Indianapolis. This makes it available for more efficient advance development because the Allison plant is at Indianapolis and Aerodynamics is at Seattle, Dayton, Ohio.

BOAC Piles Up Hours on Comets

(McGraw-Hill World News)

London-Edinburgh Overseas Airways Corp.'s six de Havilland Comet jet airliners has flown more than 1,500,000 sq. mi. and 2,445 hrs. to Feb. 14. The Comet flew 815 hrs. on BOAC's London-Edinburgh route, providing on training, development and proving.

Delivery of a seventh Comet is expected shortly. BOAC has an order now. Also, the Comet is in order now. Also, the Comet is in order now.

Airport Aid For Fiscal 1953

A total of \$68 airport construction or development projects, involving expenditures of \$19,053,585 to be financed by Civil Aeronautics Administration during fiscal 1953. Local and state project sponsors are to receive \$9,070,665 at the cost of the projects, some of which concern new airport construction, only improvement of existing facilities. Revenues of the funds will be contributed by the federal government.

On June 30, 1952, the end of the sixth year of the Federal Aid Airport Program, a total of \$83,145,975 in federal funds had been programmed.

The projects include 20 international airports, 15 conventional, 24 general airports, 51 trails, 43 hangers, and 14 secondary airports.

International projects: Oakland, Calif., Municipal Airport, \$16,144; San Francisco, \$94,585; Denver, \$50,761; Miami International, \$190,944; Atlanta Municipal, \$103,916; Chicago O'Hare Airport, \$610,000; New Orleans Municipal Airport, \$93,758; Baltimore, Frederick International, \$112,000; Boston, \$500,800; Detroit, \$140,000; Minneapolis-St. Paul, \$45,000; St. Louis Lambert Airport, \$58,312; New York International, \$400,000; Portland, Ore., International, \$112,713; Philadelphia International, \$400,000; Ft. Worth International, \$93,316; Houston Municipal, \$200,000; Seattle Boeing, \$250,000; Seattle-Tacoma International, \$141,000; San Jose, Puerto Rico, International, \$60,000.

Seek Pay Raises

(McGraw-Hill World News)

Melbourne-Australia airline pilots are shaping their strategy to get higher pay. For the first time, the pilots intend to bypass direct negotiations with the airlines and instead will utilize official government union consultation machinery. Present annual salaries range from \$4,400 for a Grade 1 captain down to \$8,750 for a probationary first officer.

United Sells Lamsa To Mexican Firm

(McGraw-Hill World News)

Mexico City—United Air Lines has sold its Mexican subsidiary, Lamsa Servis Mexicana S.A. (Lamsa) after 15 years of operation. The line was purchased at an undervalued price by Aerovias de Mexico, S.A., which will continue its incentive Mexico City Airport plans with the firm's services to central northwest Mexico. Lamsa originally was set back over

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Complexity, the Frankenstein

Are airplanes too complex?

Must the crowd march toward ever more complexity (and cost) be stopped in its tracks?

Latest reports to AVIATION WEEK reveal that more and more influential aviation people are beginning to say yes.

The general subject of complexity is slowly rising to the top among urgent priority topics for discussion. The implications of continuing toward greater complexity at the historic pace of the past ten years are slowly dawning on more of the best thinkers.

Some aspects we perceive cannot be publicized at the present stage. However, AVIATION WEEK's West Coast editor, after reviewing several individuals concerning the confidential design meetings recently held by the Institute of the Aeronautical Sciences, says:

"Our aircraft designers definitely are thinking in terms of lighter, more simplified planes for the future. They are worried about the trend toward heavier and more complicated aircraft. Future design will be lighter, leaner, as well as lighter."

This subject was competently publicized recently by E. H. Heinemann, chief engineer of the El Segundo division of Douglas Aircraft Co. His address was quoted on this page, Page 38.

Quoted further by AVIATION WEEK, Mr. Heinemann reveals that he is no pessimist on the possibilities of slowing the overwhelming flood of complexity. He writes further:

"With respect to what can be done to attack the problem of complexity, it is my personal opinion that the best progress can be made with an educational campaign."

"Toward that end we have been giving numerous presentations to government, to government, and engineers as personnel of the service, and find they are all generally in agreement."

"Our strongest recommendation to the service is that every requirement be analyzed especially for the airplane involved, and that each requirement be made to stand upon its own feet whether or not there is a military requirement, general doctrine, or specification."

We asked Mr. Heinemann: "To your knowledge, is any intensive effort by anyone to top rebel industry or government circles being expended on the subject of aircraft complexity?"

Mr. Heinemann says:

"There is a great deal of concern in top rebel industry and government circles about this problem, but most top level personnel are not sufficiently acquainted with detail design problems to know how to integrate the more complicated requirements and to take corrective action. This again is where education, originating from the experienced engineers, is essential. The Aircraft Industries Association is also taking interest."

"I believe that most top industry and military officials feel the way I do, especially the older and more experi-

enced ones; perhaps, however, not so strongly. In most cases, top management is so deeply involved in management problems that they have little time to devote to this subject."

Mr. Heinemann, at our invitation, told how he has been AVIATION WEEK (and the business press generally) can be of aid in promoting greater interest in the subject of complexity.

"I am sure you can be of great assistance, especially along the lines of educating all concerned with present aircraft. In this connection, while rather idealistic, perhaps one of the best ways would be to give praise for simple, efficient designs rather than being too critical of the less successful designs."

It is also Mr. Heinemann's belief that if the individuals concerned could be persuaded to think in terms of long-range planning, and obtaining an end result rather than permitting each activity to battle for its individual requirements, the situation could be improved.

"After all," he says, "an airplane is the most highly complicated mechanism most of us know about, and its success can only be measured in the end result."

The obstacles to top simplicity are seen insurmountable and the reason for more complexity are many and powerful. But if we permit this Frankenstein of complexity to continue work at its current gliding, useless rate, it will slowly overwhelm us to ineptness.

Spots Before Our Eyes

We don't know what "flying saucers" are.

For two years everyone on our staff has followed instructions to ask penetrating questions of the highest aviation officials in government and industry. We have failed to find a hint that one of them knew any more than we did.

We do believe President Truman and several defense officials are being truthful when they say these floating objects were no product of our defense industry.

It is the most baffling news story we have ever tried to nail down. And no theory we have had satisfies us.

The Air Force finally got around to having a full-day news conference, dignifying the subject. That was a long step from its earlier attitude that people who asked about these silly things were not completely equipped in the belly.

Despite all of the learned comments that were discussed at the official press conference, it seems significant that these military intelligence experts and scientists broke down and admitted that 22 of their findings and research still leave 25% of the sightings unexplained!

So as far as solving the mystery, we are about where we were before the press was called in. The only progress that appears to be evident is that more people, and important people, are acting as though such things might cost over \$1.

This is all to the good. No one is running about the highways that can be explained. Let's discard these and get down to business on the elusive 25%.

—Robert H. Wood

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Photo courtesy of North American Aviation Inc., Commercial Aircraft, Los Angeles-Calif.

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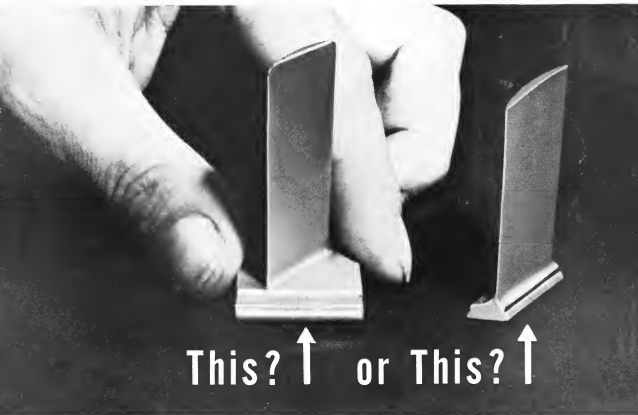
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Which One Will Save a Million Dollars?



Three years in the making, the fabricated jet engine compressor stator blade (left) promises to save the armed forces not just one million, but millions of dollars annually in jet engine costs, compared with the forged blade (right). This new G-E development will cut manufacturing cost in half and save over a third in critical materials. Military approval has been received for the use of fabricated blades in the General Electric J47-GE-23 which powers the Boeing B-47 Stratojet bomber. And G.E., through the United States Air Force, is sharing the process with other turbojet manufacturers.

The blades are rolled in long strips, contoured to the proper air foil, and cut to desired length. Each blade is then welded into a separate base which fills the same

area as the "blade ring" used with forged blades. Thus the ring and an expensive manufacturing and assembly process have been eliminated.

Endurance tests on two engines equipped with the fabricated blades proved them just as efficient as forged blades. The base provides greater resistance to vibration due to uneven airflow through the compressor. Damage caused by foreign objects entering the compressor is minimized because the new blade is fastened much more strongly to the casing.

A product of G-E research at the Thomson Laboratory in Lynn, Mass., this new method of manufacturing stator blades is another of the many ways in which G.E.'s constant pioneering contributes to the advancement of aviation. General Electric, Schenectady 5, N.Y.

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